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High Fidelity Modeling of Field-Reversed Configuration (FRC) Thrusters

*Space Propulsion and Power Portfolio Review
28 September – 02 October 2015*



**Justin Koo
AFRL/RQRS**



Overview



- **Background**
 - Branch-level strategy
 - Status of experimental efforts
- **Project Description**
 - Goals/Objectives/Technical Challenge/Approach
- **Progress Update**
 - Low fidelity (Hugrass) status
 - High fidelity (Multifluid) status
 - 2-fluid FRC simulation
- **Conclusions**



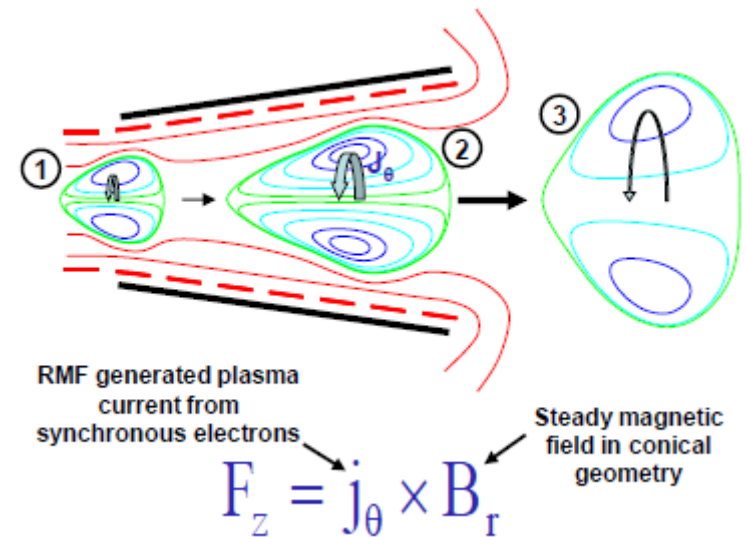
Background



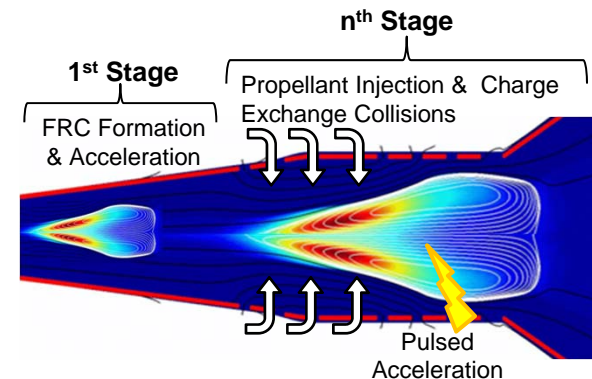
FRC propulsion



- **Field Reversed Configuration (FRC)** is offshoot of fusion research
 - Ionization by rotating B-field
 - Pulsed inductive $j \times B$ acceleration
 - Magnetically insulated, plasmoid accelerated downstream
- **Key attributes**
 - Very low mass (estimate ~1-2 kg/kW)
 - Efficiency comparable to or higher than Hall thrusters (predicted)
 - Operates on diverse propellants; potential for multi-mode application
 - Pulsed operation provides near-constant efficiency over wide power range



Multi-Stage Neutral Entrainment

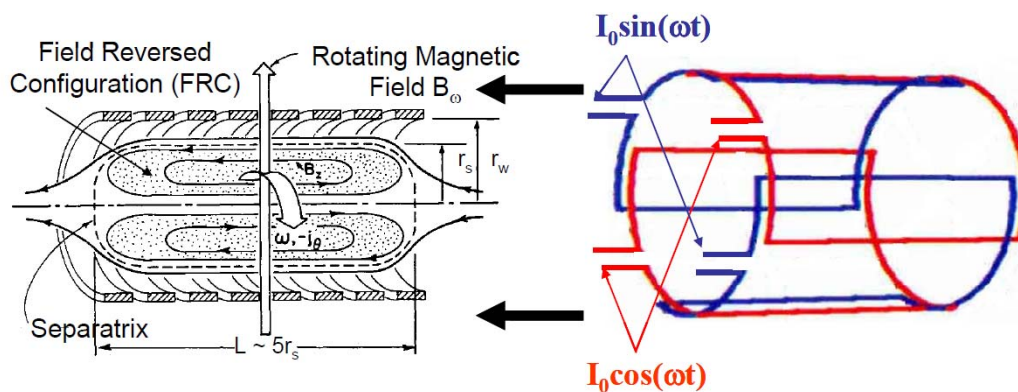
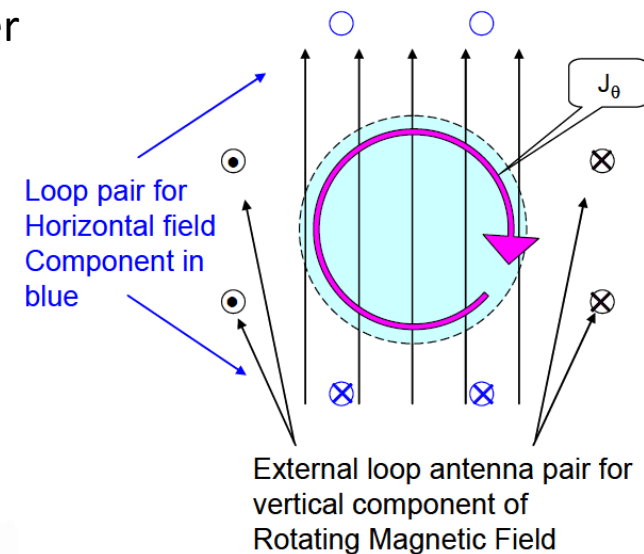




Rotating Magnetic Field (RMF)



Electrodeless Lorentz Force (ELF)
Rotating Magnetic Field (RMF)
thruster



RMF FRC thruster design appears well suited for low power operation; different set of computational challenges than theta-pinch RMF simulation

Slough, J, Kirtley, D and Weber T, **Pulsed Plasmoid Propulsion: The ELF Thruster**, IEPC-2009-265, 31th International Electric Propulsion Conference, Ann Arbor, Michigan, September 20-24, 2009.



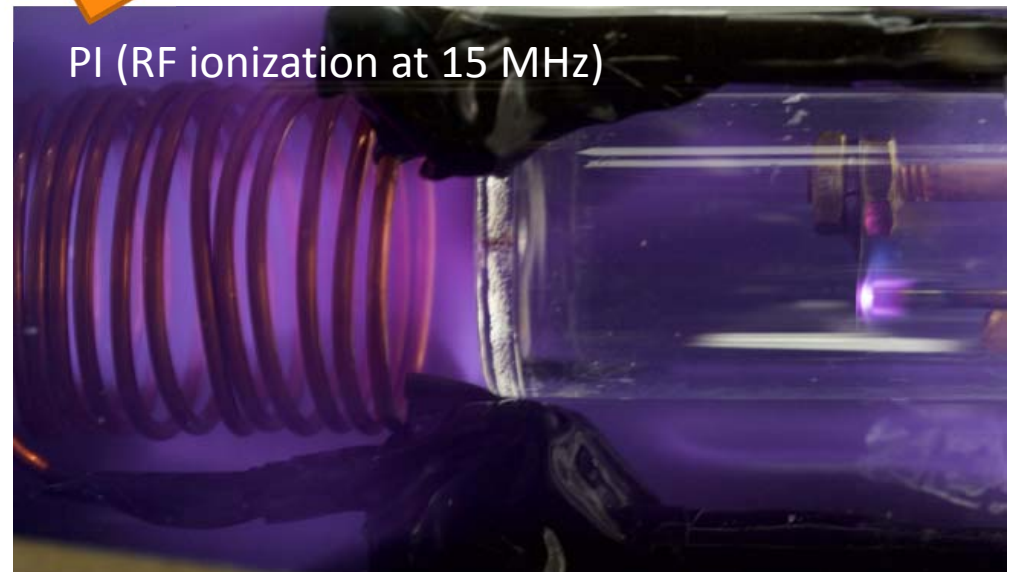
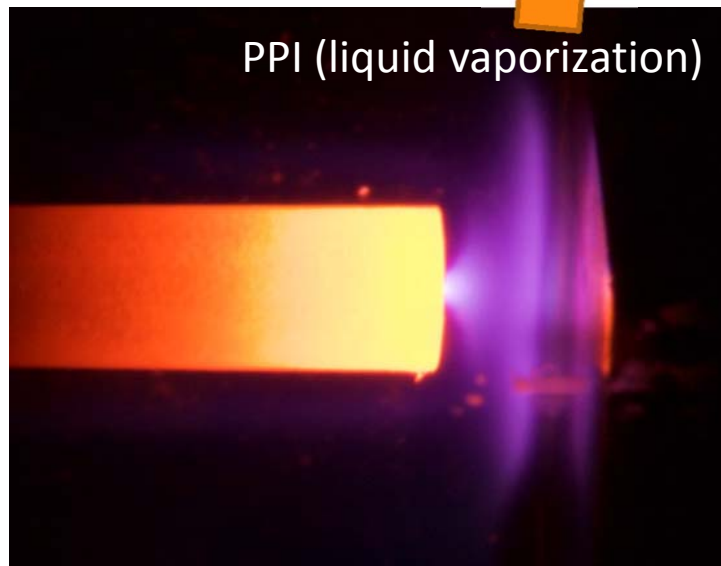
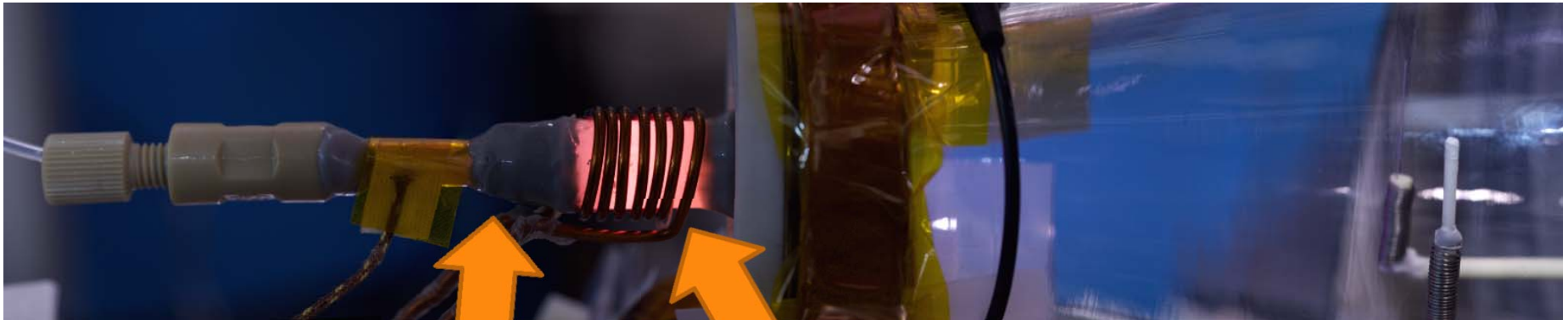
Branch-level FRC development strategy



- **In-house**
 - Computational 6.1 program to develop numerical tools (this program)
 - Experimental 6.2 program to generate validation data
 - Briefly attempted theta-pinch configuration, failed to make sufficient progress (limited amount of data)
 - Switched to rotating magnetic field (RMF) configuration
 - Simultaneous development of PI/PPI systems
- **External**
 - Leveraging SBIRs to develop engineering-level FRC models for testing
 - Testing thrusters at MSNW LLC (Redmond, WA) and AFRL/RQRS (Edwards AFB, CA)

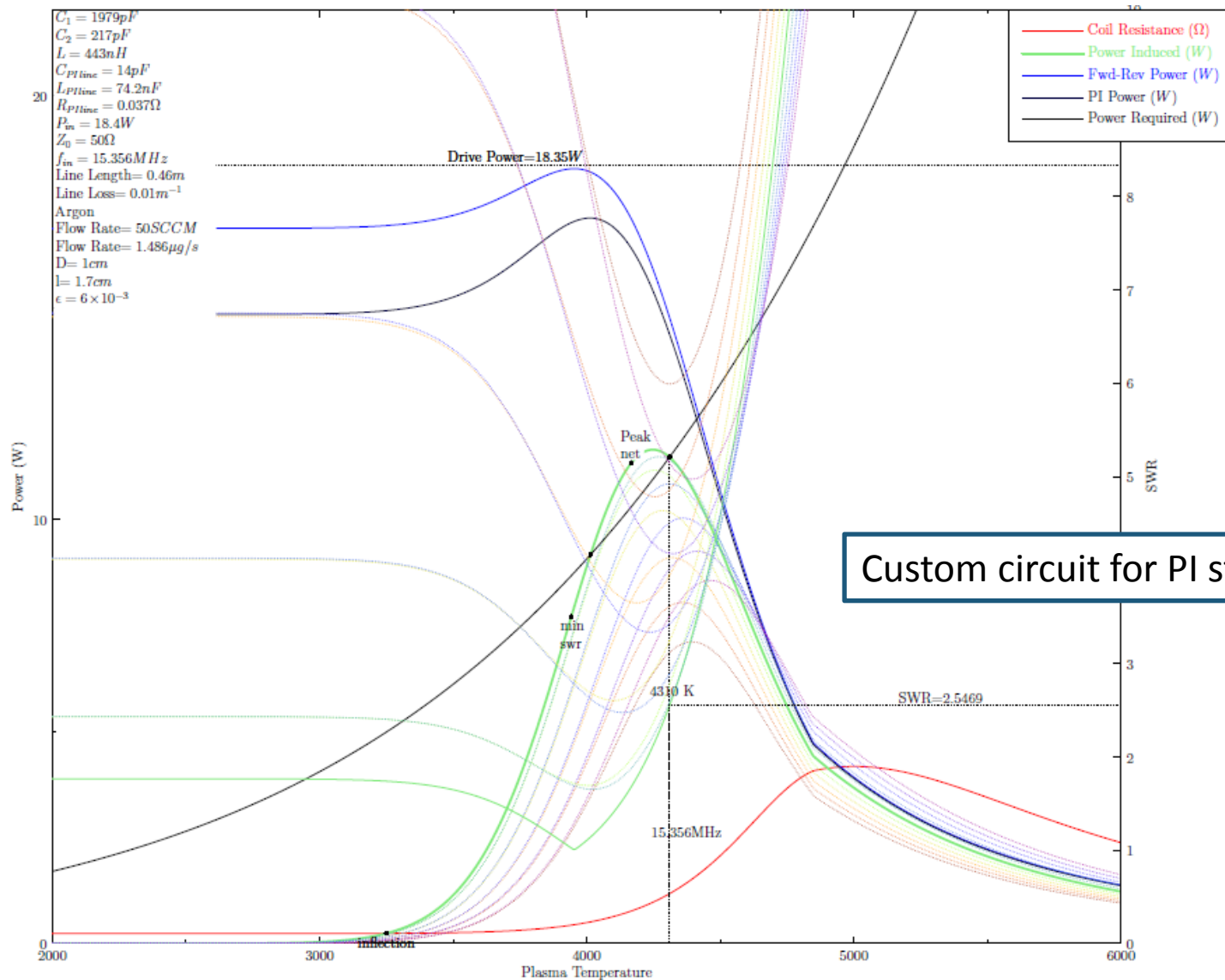


Experimental efforts (1)





Experimental efforts (2)





Project Description



Justification



- **Some computational models exist**
 - Neutral flow dynamics
 - Fully-ionized FRC translation
 - Collisional kinetic of charge-exchange
- **The right integration of numerical models does not exist**
 - Partially ionized magnetized plasmas are difficult to simulate
 - Modern diagnostics include internal/external probes and spectroscopy
 - Simulation of realistic FRCs requires specific physic modules such as coil-plasma interaction models

Fundamental question is how to scale FRCs to run efficiently at low power levels – difficult/impossible without simulation



Goals



- **Pre-ionization** – need to generate some amount of seed plasma; for multi-plasmoid operation, first pulse is the most difficult; possible to consider many different configurations (e.g. liquid phase propellants)
 - Very specific to configuration
 - Introduction of multiphase and chemistry greatly complicate physics
- **Formation** – go from seed plasma to strongly ionized closed-field plasmoid; kinetic effects expected in first stage; interaction between coil/plasma drive process
- **Translation** – plasma remain almost fully ionized; interested primarily in collisional effects as plasma impact neutral background
 - Fairly straightforward to model with existing fluid tools (high enough collisionality for Maxwellian assumption)

Focus on Formation → develop hierarchy of tools to understand plasmoid formation, test spacecraft / integration and design new thrusters



Objectives

- **Multi-scale / multi-physics capability** – emphasis is on high order methods and consistent collision modules
- **Formation and acceleration dynamics** – validation with experimental test campaign
- **FRC stability and turbulence onset** – fundamental physics of current sheets in plasma
- **Collisional-radiative characteristics** – detailed spectral signatures to reduce need to perturb plasma

Well-developed software framework will support not only FRC simulation, but also development of future HET / plume / general partially ionized, magnetized plasma devices



Technical Challenges



- **Field/plasma model** – Plasmas can have very complex phase space configurations and self-induced electromagnetic forces. Fundamental multiscale problem since electrostatic forces are relevant in this problem and force very small timescales.
- **Collisional Physics** – kinetic of ionization (and especially energy loss) are very sensitive to details of electron energy distribution function
- **Multi-Scale Effects** – diffusive behavior in region of separatrix is not modeled correctly in collisionless code – artificial diffusion makes computational phenomena look real.
- **System Complexity** – realistic simulations need to couple circuit models to plasma, represent three-dimensional / end effects, address radiation loss to impurities, etc.



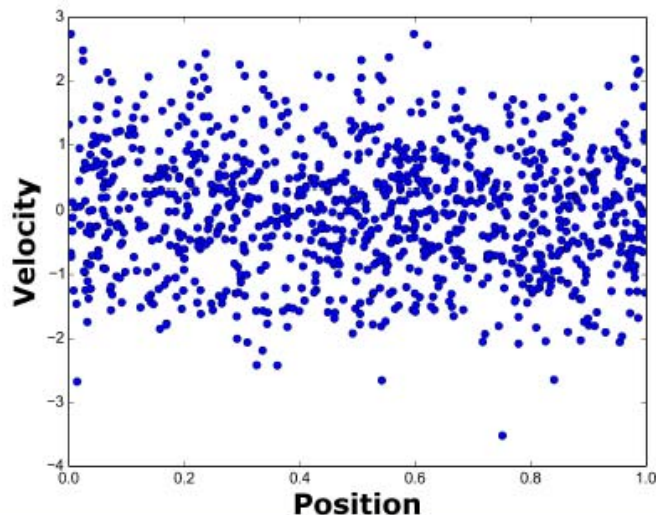
Approach



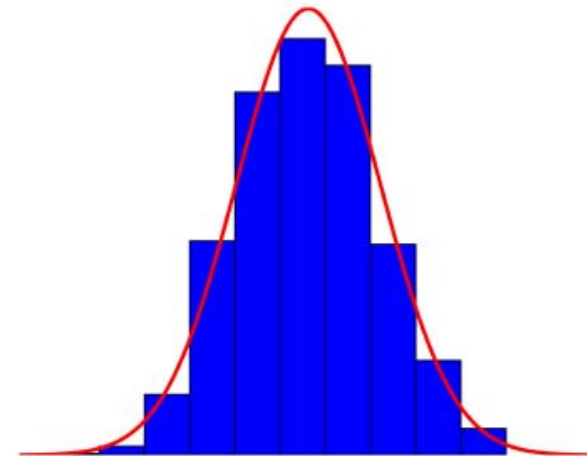
- **Hierarchy of Plasma Models**
 - Continuum: Hugrass – Ideal MHD – Hall/Resistive MHD – MF
 - Kinetic: Explicit – Implicit Particle-in-cell (PIC)
- **Collisional Physics** – Consistent derivations from underlying cross-sections for both C-R and MF coupling terms
- **Code acceleration** – Development of physics modules in next-generation simulation framework at AFRL/RQRS; leveraging both software (algorithmic) and hardware (GPU) acceleration



Plasma Description



- 3-Dimensions + 3-Velocities
- Evolve the particles position and velocity
- e.g. Particle-In-Cell models



- Ensemble average of particles distribution, $f_s(\mathbf{x}, \mathbf{v}, t)$
- Evolve the distribution function
- e.g. Vlasov-Maxwell models



Kinetic (non-Maxwellian)

- The Boltzmann eqn:

$$\frac{\partial f_s}{\partial t} + \mathbf{v} \cdot \frac{\partial f_s}{\partial \mathbf{x}} + \frac{q_s}{m_s} (\mathbf{E} + \mathbf{v} \times \mathbf{B}) \cdot \frac{\partial f_s}{\partial \mathbf{v}} = \frac{\partial f_s}{\partial t} \Big|_c$$

- Take the 0th, 1st, 2nd moments of the Boltzmann Eqn.

$$m_s \int \mathbf{v}^n \frac{\partial f_s}{\partial t} d\mathbf{v} + m_s \int \mathbf{v}^{n+1} \cdot \frac{\partial f_s}{\partial \mathbf{x}} d\mathbf{v} + q_s \int \mathbf{v}^n (\mathbf{E} + \mathbf{v} \times \mathbf{B}) \cdot \frac{\partial f_s}{\partial \mathbf{v}} d\mathbf{v} = m_s \int \mathbf{v}^n \frac{\partial f_s}{\partial t} \Big|_c d\mathbf{v}$$

- Each moment of the Boltzmann eqn gives an equation for the moment variable, and introduces the next higher moment variable
- This process can go on indefinitely

Can implement Boltzmann equation multiple ways:

- **PIC** (high dimensionality – 3D3V; high statistical noise)
- **Vlasov** (low dimensionality – 2D2V hero runs; smooth solution)

Leverage 6.2 program to provide codes and expertise



Moments of the distribution

- Modeling each particle velocity and position is not practical.
- Instead an average is performed to give a statistical description.
- Calculate the number of particles per unit volume having approximately the velocity \mathbf{v} near the position \mathbf{x} and at time t , distribution function $f(\mathbf{v}, \mathbf{x}, t)$

$$\rho_s = m_s \int f_s(\mathbf{v}) d\mathbf{v}$$

$$\rho_s \mathbf{u}_s = m_s \int \mathbf{v} f_s(\mathbf{v}) d\mathbf{v}$$

$$\mathbb{P}_s = \mathbf{P}_s = m_s \int \mathbf{w} \mathbf{w} f_s(\mathbf{v}) d\mathbf{v}, \quad p_s = \frac{1}{3} m_s \int w^2 f_s(\mathbf{v}) d\mathbf{v}$$

$$\mathbf{H}_s = m_s \int \mathbf{w} \mathbf{w} \mathbf{w} f_s(\mathbf{v}) d\mathbf{v}, \quad \mathbf{h}_s = \frac{1}{2} m_s \int w^2 \mathbf{w} f_s(\mathbf{v}) d\mathbf{v}$$

$$\mathbf{w} = \mathbf{v} - \mathbf{u}_s$$

No assumptions on shape of distribution function yet....still a general fluid!



Multifluid (MF) Plasma Model



Fluid Description (3 equations for each specie)

Mass $\frac{\partial \rho_s}{\partial t} + \nabla \cdot (\rho_s \mathbf{u}_s) = S_s$

Mom. $\frac{\partial \rho_s \mathbf{u}_s}{\partial t} + \nabla \cdot (\rho_s \mathbf{u}_s \mathbf{u}_s + p_s \overleftrightarrow{\mathbf{I}}) = \frac{\rho_s q_s}{m_s} (\mathbf{E} + \mathbf{u}_s \times \mathbf{B}) + \sum_r \mathbf{R}_{rs} - \nabla \cdot \overleftrightarrow{\mathbf{\Pi}}_s$

Energy $\frac{\partial \varepsilon_s}{\partial t} + \nabla \cdot ((\varepsilon_s + p_s) \mathbf{u}_s) = \left(\frac{\rho_s q_s}{m_s} \mathbf{E} + \sum_r \mathbf{R}_{rs} \right) \cdot \mathbf{u}_s + \overleftrightarrow{\mathbf{\Pi}}_s : \nabla \mathbf{u}_s - \nabla \cdot \mathbf{q}_s + \sum_r Q_{rs}$

Closure through LTE

Interspecies collisional terms

ELECTROMAGNETIC FIELDS:

$$\frac{\partial \mathbf{B}}{\partial t} + \nabla \times \mathbf{E} = 0, \quad \nabla \cdot \mathbf{B} = 0$$

$$\frac{1}{c^2} \frac{\partial \mathbf{E}}{\partial t} - \nabla \times \mathbf{B} = -\mu_o \sum_s \frac{q_s}{m_s} \rho_s \mathbf{u}_s, \quad \epsilon_o \nabla \cdot \mathbf{E} = \sum_s \frac{q_s}{m_s} \rho_s$$



Further Simplification

- If there are no neutral species, sum all charged species to get a single charged “fluid” equation
- Connect with electromagnetism through Faraday’s law and suitable version of Ohm’s Law

(Faraday’s Law)
$$\frac{\partial \mathbf{B}}{\partial t} + \nabla \times \mathbf{E} = 0$$

MHD variants of Ohm’s Law

Ideal	$0 = \vec{E} + \vec{u} \times \vec{B}$	Magnetic field frozen in flow (no diffusion)
Resistive	$\vec{j} = \sigma(\vec{E} + \vec{u} \times \vec{B})$	Magnetic field can diffuse into plasma
Hall	$\vec{j} = \bar{\sigma} \vec{E}$	Incorporates Hall currents
Resistive Hall	$\vec{j} = \bar{\sigma}(\vec{E} + \vec{u} \times \vec{B})$	Hall currents and magnetic field diffusion

- Remove fluid equations altogether and simply represent plasma as finite resistivity affecting Maxwell’s equations – Hugrass RMF model

Need all models in same framework to facilitate simulation at multiply levels of fidelity → can identify role of missing physics



Detailed C-R

- **Require spectral signatures to interrogate many regions of the plasma without significantly perturbing plasma**
 - Probes don't function well in RMF environment
- **Leveraging 6.2 and 6.1 work to access detailed Ar models**
 - Based on theoretical and experimental databases from NIST, LXCAT, etc.
 - 6.2 work supports detailed model development / 6.1 work has supported intelligent model reduction for computational tractability
- **Groundwork has been laid to develop self-consistent inelastic transport coefficients from same cross-section databases**



Progress Update



Numerical Model Development



- **Tackling both sides of plasma hierarchy**
 - Low-fidelity Hugrass model
 - Quick turnaround so more useful for interfacing with experimentalists
 - Potential low-order model for time-parallel acceleration strategy
 - High-fidelity Multifluid MHD model
 - Bringing SoA multifluid capability to FRC modeling
 - High order DG formulation
 - Working on extension to three-fluid
 - Maintaining consistency with C-R rates
- **Computational machinery for high-fidelity will be used to build MHD (single fluid) system**



Hugrass model

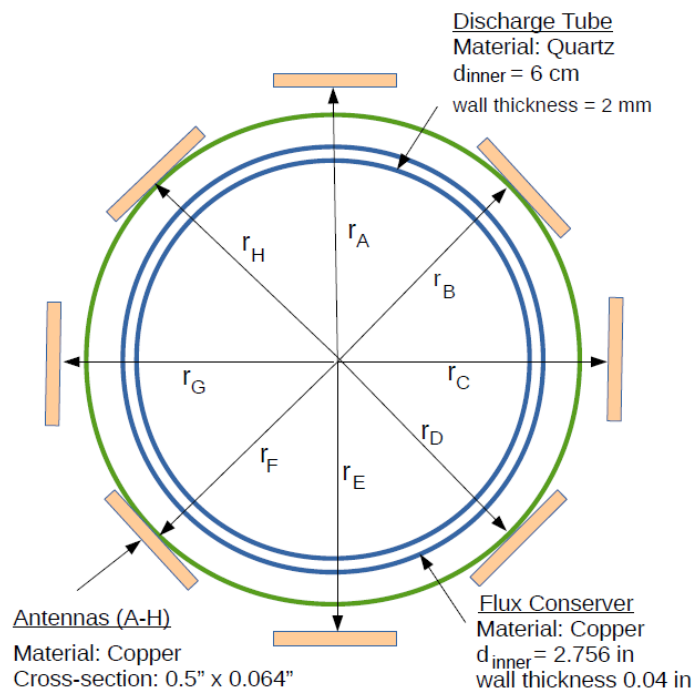
- Implementation of Maxwell's equations in non-vacuum
- Based on numerical model originally published in 1981 by Hugrass (J. Plasma Physics, **26**, 455-464)
- Radial-azimuthal code (infinitely long cylinder)
- Coupled equations for axial magnetic field and axial component of magnetic vector potential; solved spectrally

$$\begin{aligned}\frac{\partial A_z}{\partial t} &= \frac{\eta}{\mu_0} \nabla^2 A_z + \frac{1}{n_e e \mu_0 r} \left[\left(\frac{\partial A_z}{\partial r} \right) \left(\frac{\partial B_z}{\partial \theta} \right) - \left(\frac{\partial A_z}{\partial \theta} \right) \left(\frac{\partial B_z}{\partial r} \right) \right] \\ \frac{\partial B_z}{\partial t} &= \frac{\eta}{\mu_0} \nabla^2 B_z + \frac{1}{n_e e \mu_0 r} \left[\left(\frac{\partial A_z}{\partial \theta} \right) \frac{\partial}{\partial r} \nabla^2 A_z - \left(\frac{\partial A_z}{\partial r} \right) \frac{\partial}{\partial \theta} \nabla^2 A_z \right]\end{aligned}$$

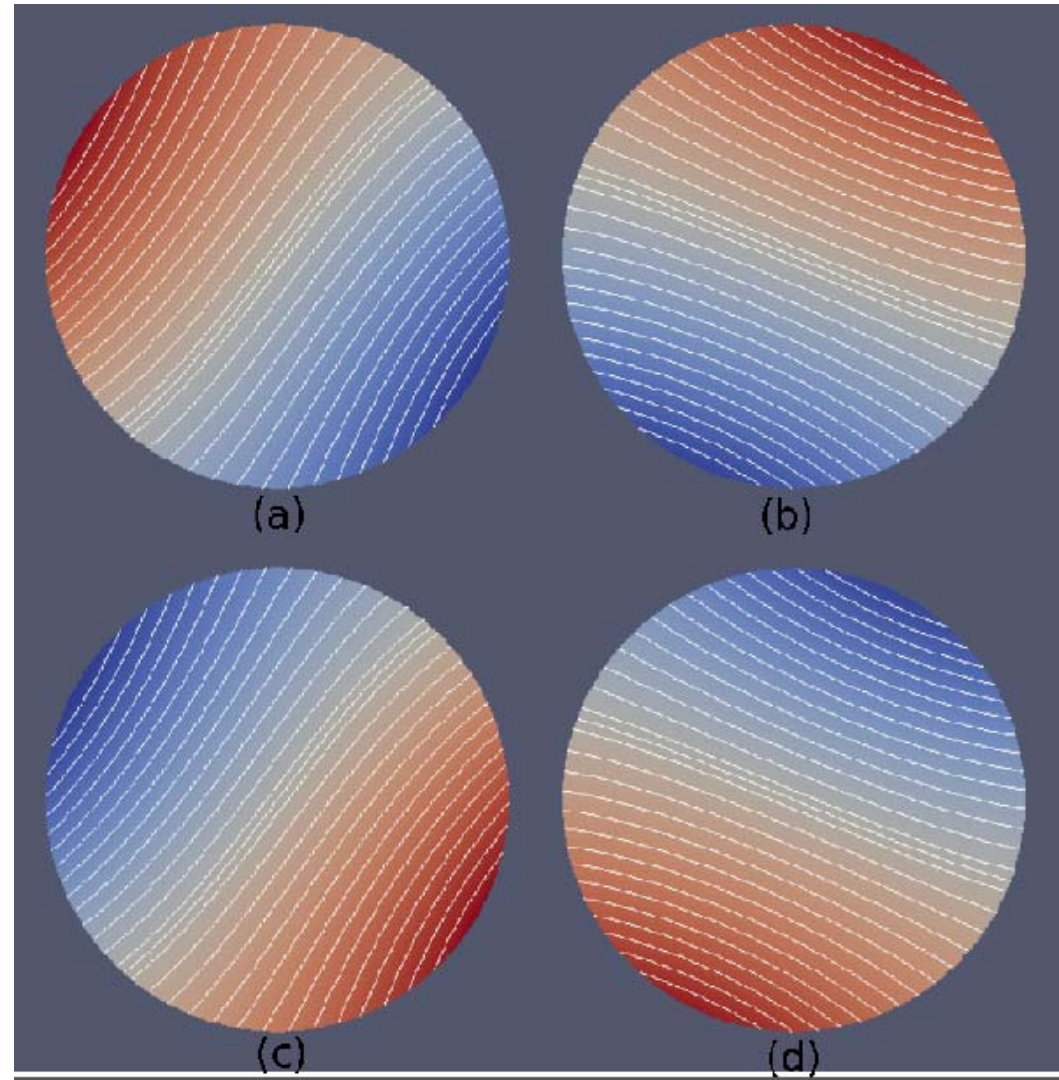
- No self-consistent plasma transport – can still define spatial/temporal dependence of resistivity (η)



Vacuum field of RMF FRC



$$\begin{aligned} r_{A,B} &= -l(t) \cos(\omega_{RMF}t + \varphi) \\ r_{C,D} &= l(t) \sin(\omega_{RMF}t) \\ r_{E,F} &= l(t) \cos(\omega_{RMF}t + \varphi) \\ r_{G,H} &= -l(t) \sin(\omega_{RMF}t) \end{aligned}$$





Development of MFPM



IDEAL MHD MODEL IS VALID WHEN:

- High collisionality, $\tau_{ii}/\tau \ll 1$
- Small Larmor radius, $r_{Li}/L \ll 1$
- Low Resistivity, $\left(\frac{m_e}{m_i}\right)^{1/2} \left(\frac{r_{Li}}{L}\right)^2 \frac{\tau}{\tau_{ii}} \ll 1$

MULTI-FLUID PLASMA MODEL

- Less computationally expensive than kinetic models
- Multi-fluid effects become relevant at small spacial and temporal scales
- Finite electron mass and speed-of-light effects are included
- There is charge separation is modeled
- Displacement current effects are resolved in the MFPM



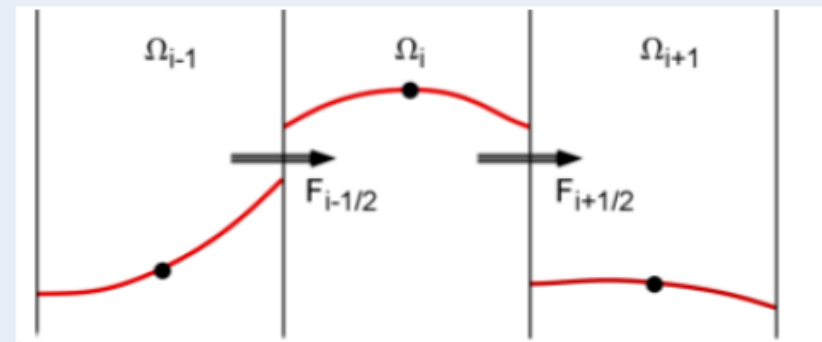
Discontinuous Galerkin (1)

- Conservation Laws are given by $\frac{\partial \mathbf{Q}}{\partial t} + \frac{\partial \mathbf{F}}{\partial x} = \mathbf{S}$
- Multiply by basis functions and integrate over the volume

$$\int_{\Omega} v_h \frac{\partial \mathbf{Q}}{\partial t} dV + \oint_{\partial \Omega} v_h \mathbf{F} \cdot d\mathbf{A} - \int_{\Omega} \mathbf{F} \cdot \nabla v_h dV = \int_{\Omega} v_h \mathbf{S} dV$$

- That is an ODE of the form $\frac{d\mathbf{Q}_h}{dt} = \mathcal{L}_r(\mathbf{Q})$, solved using 2nd, 3rd or 4th order TVD Runge-Kutta

Spatial order of 2nd-16th is common.





Discontinuous Galerkin (2)



- Riemann problems are solved at each interface to compute fluxes
- The source of dissipation / dispersion depends on the Riemann solver
- Variables are allowed to be discontinuous at the cell interfaces

Advantages

- Method is conservative
- Compact stencil for flux evaluation (easily parallelizable)
- Provides good flux-source coupling (no source splitting is needed)
- Resolves fast oscillations (e.g. plasma frequency) and large gradients (e.g. shocks)

Disadvantages

- Explicit Runge-Kutta Discontinuous Galerkin CFL limit is $< 1/(2p-1)$ where high order schemes have higher p



Multiscale issues



CHARACTERISTIC SPEEDS:

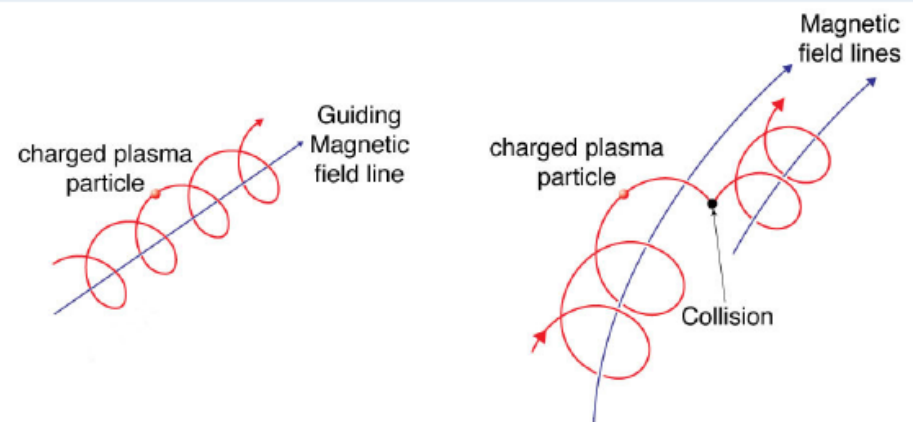
$$v_{cs} = \sqrt{\gamma \frac{P_s}{\rho_s}}, \quad c$$

CHARACTERISTIC FREQUENCIES:

$$\omega_{ps} = \sqrt{\frac{n_s q_s^2}{\epsilon_0 m_s}},$$

$$\omega_{cs} = \frac{q_s B}{m_s}$$

$$\nu_{sr} = \frac{4\sqrt{\pi} e^4 Z^4 n_r \log \Lambda}{3\sqrt{m_s} T_s^{3/2}}$$



- Example lab plasma (seconds):

$$\frac{1}{\omega_{pe}} = 10^{-14}, \frac{L}{c} = 10^{-9}, \frac{1}{\omega_{ci}} = 10^{-8}, \frac{L}{v_{ci}} = 10^{-5}$$

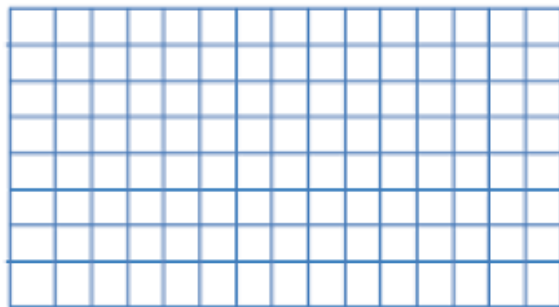


Availability



STRUCTURED DG:

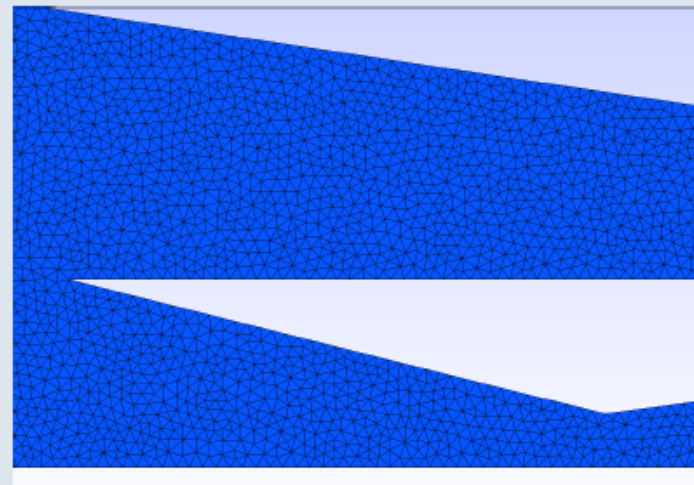
- WARPX code developed at Washington
- Validated and with proven results
- Can be used to verify the unstructured case



Structured Grid

UNSTRUCTURED DG:

- Currently being developed at AFRL
- For complex geometries/experiments
- Will include different physics modules

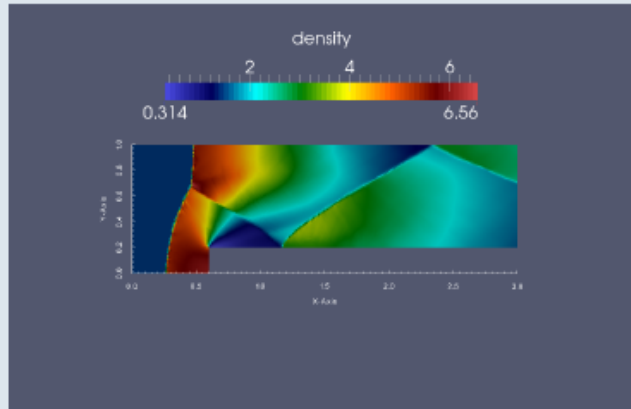




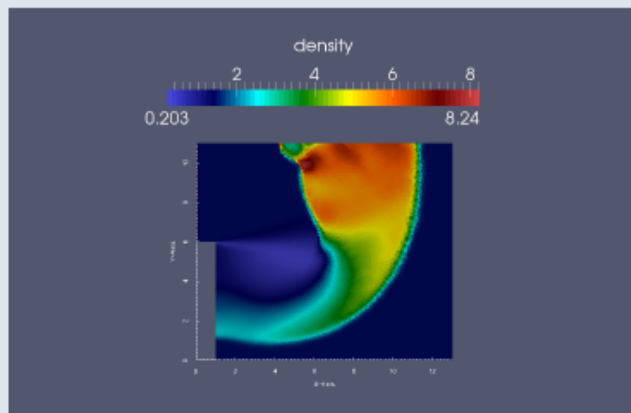
Verification



FORWARD FACING STEP



BACKWARD FACING STEP

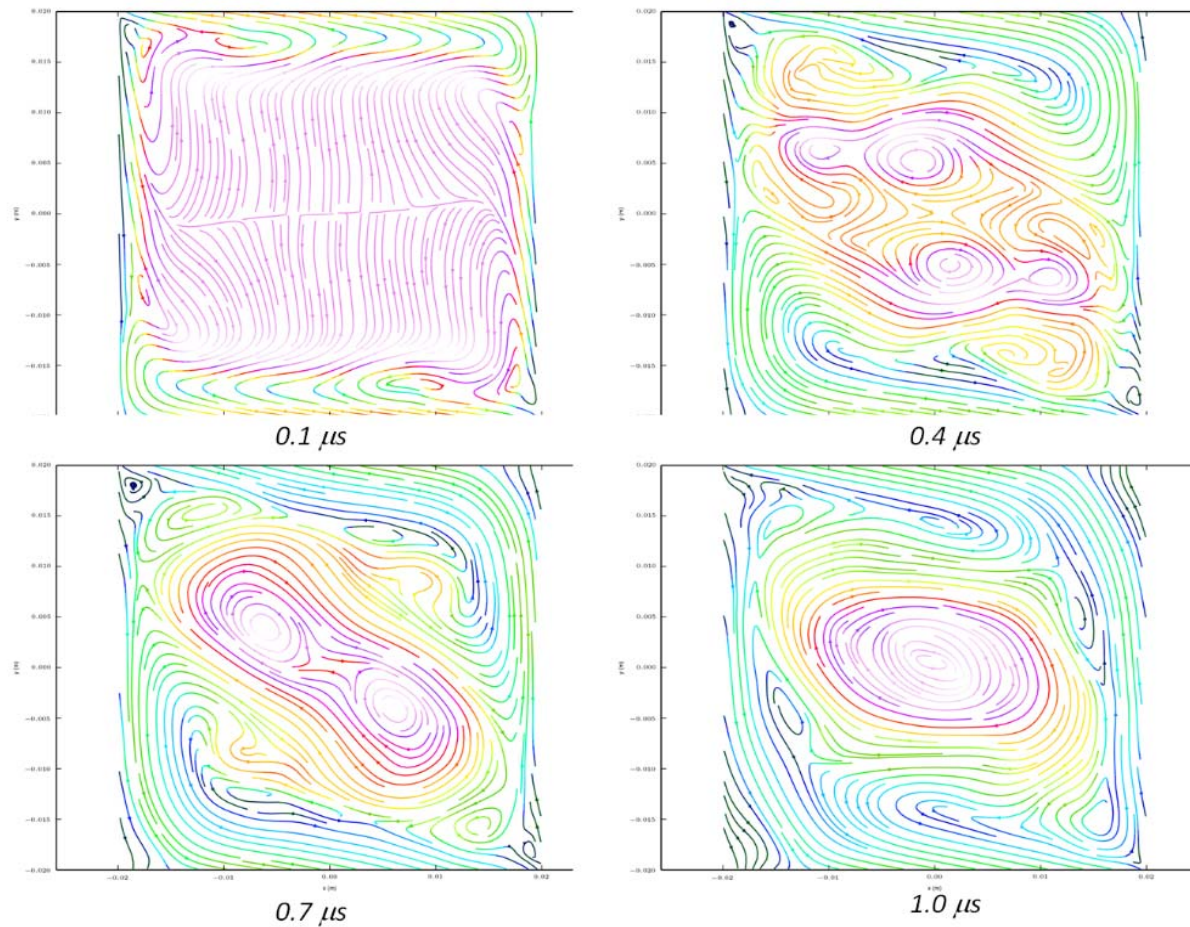


IMPLEMENTATION PIECES

- Unstructured grid readers
- Unstructured data structures
- PETCs library for data manipulation and time integration
- XML input file reader
- Quadrature rules integration
- Different Riemann solvers for numerical flux (e.g. Roe, HLL, HLLE)
- Flux Limiters for shock capturing
- Initial/boundary conditions



RMF startup (1)

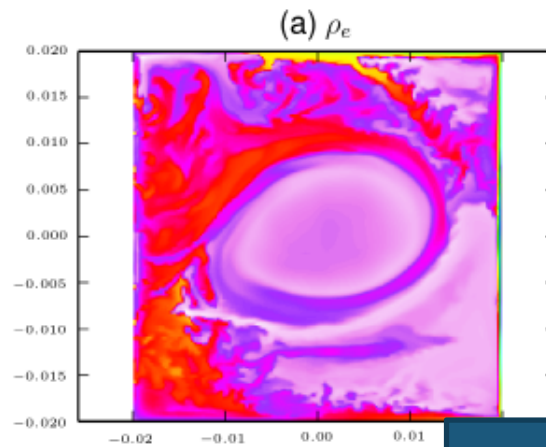


Exploring collisionless limit for efficient FRC formation

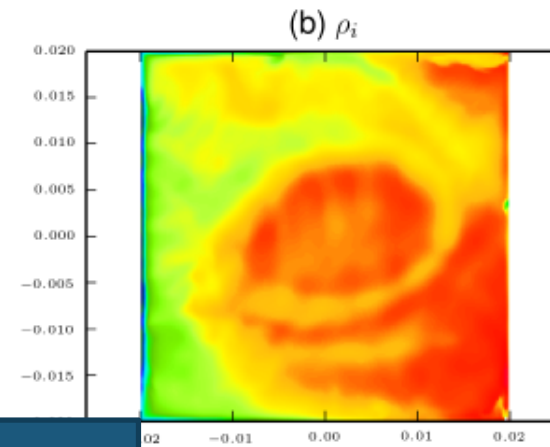


RMF startup (2)

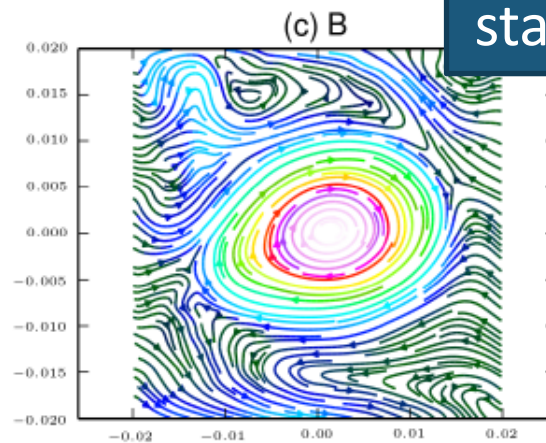
Electron mass density



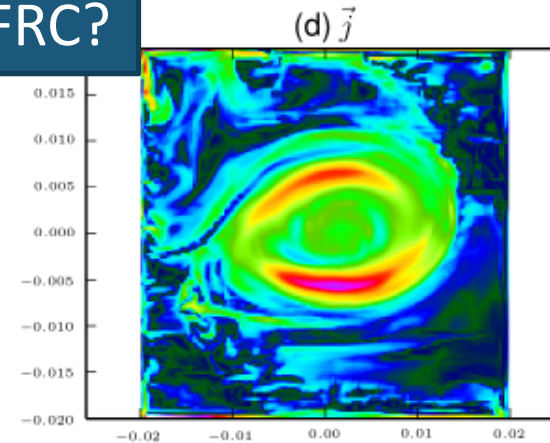
Ion mass density



4x later,
stable FRC?



Magnetic field



Current



Next Steps



- **Numerical**
 - Develop/run continuum plasma hierarchy
 - Run kinetic plasma codes
 - Integrate with collisional physics
 - Code acceleration
- **Physics**
 - FRC formation validation
 - Magnetic reconnection
 - Instability modeling



Conclusions



Project Highlights

- **Leverages 6.1 and 6.2 efforts**
 - Explicit Particle-In-Cell (PIC) and implicit PIC development for Hall Effect Thrusters (6.2, AFRL/RQRS)
 - C-R model reduction (6.1, Luginsland/Marshall)
 - Algorithm acceleration (6.1, Farhoo)
- **Integration of plasma modules into framework code provides tools for investigation of numerous plasma phenomena**
 - Adapt Multifluid description to study electrostatic anomalous electron transport due to fluid instabilities
 - Use Multifluid / MHD description as preconditioner for implicit PIC
 - Detailed C-R necessary to evaluate utility of reduced models (such as Quasi-Steady State (QSS) approximation)

Common simulation framework has potential to greatly increase transition opportunities by decreasing integration time



Summary



- **Research plan directly supports R&D objectives for advanced plasma propulsion (i.e. FRCs)**
- **Progress made in various areas:**
 - Moving towards *consistent* plasma hierarchy
 - Integrating additional physics modules for real engineering utility
 - Need more algorithm work (accuracy/stiffness)
- **Fundamental work, can be applied to other areas**
- **Challenging, *exciting* project....**
- **....but off to a slow start**